



# Influence of boundary thermal fluctuations on event coordinate determination in a Time-projection chamber

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# Nuclotron-based Ion Collider fAcility

NICA – heavy-ion accelerator complex in the framework of Megascience project

**NICA project objective:** Study of the phase diagram of nuclear matter

#### Accelerator complex:

- Nuclotron basic synchrotron for ion acceleration;
- NICA Collider collisions on counter-punches;
- Multi-Purpose Detector the main NICA detector.



#### https://nica.jinr.ru/ru/

#### Energy range: 4 - 11 GeV/nucleon.

#### Infrastructure of the NICA Accelerator Complex

### Time-projection chamber

#### **Time-projection chamber –** MPD

primary sub-detector

#### Principle of work:

- A nuclear reaction product ionizes the argon-methane mixture. This process creates an electron track;
- Electrons drift in electric flied toward detectors located at both ends of the cylindrical chamber;
- Thermal stabilization of the internal gas volume ensures constant electron drift velocity and accurate particle identification



**Goal:** explore the influence of thermal stabilization on the precision of TPC

#### **Objectives**:

- Estimate the temperature inhomogeneity degree at the boundary of the working gas volume in TPC;
- Determine the influence of temperature inhomogeneity on event detection accuracy.

# External thermal screen

**External thermal screen** – set of aluminum plates with water pipes bounding the working gas volume.

**Experiment**: the panel is placed in an insulated box and brought to equilibrium, then cold water is 'instantaneously' supplied  $T_{water} = T_{room} - 12.6 \text{ K}$ , 1 m/s



The difference in the curves slope is explained by different quality of thermal contact between the pipe and the radiator - the resistance of the adhesive layer

### Inhomogeneity on the thermal screen

Screen is being heated from outside by Time-of-flight detector. Temperature spatial distribution is expected to be non-uniform due to difference in adhesive thermal resistance and screen construction

We consider possible temperature amplitude from 0.1 to 2.5 K

 $F(x, y, z) = T_a \cdot \sin\left(k_{long} \cdot \frac{\pi}{L_{gasTPC}} z\right) \cdot \sin(k_{rad} \cdot \operatorname{arctg}(y, x))$ 



Temperature at gas volume boundaries



### The model & the mesh

Navier-Stokes & heat transfer equation:

$$\begin{cases} \nabla \cdot \vec{u} = 0 \\ \rho_0 \frac{\partial \vec{u}}{\partial t} + \rho_0 (\vec{u} \cdot \nabla) \vec{u} = -\nabla \left( p + \mu_0 \cdot \nabla \vec{u} \right) + \rho_0 \beta \left( T_0 \right) \left( T - T_0 \right) \vec{g} \\ \rho_0 C_{p0} \left( \frac{\partial T}{\partial t} + (\vec{u} \cdot \nabla) T \right) = \nabla \cdot \lambda_0 \nabla T \end{cases}$$

 $\vec{u}$  – velocity;

ρ – density;

 $\mu$  – dynamic viscosity;

*p* – pressure;

*t* – time;

 $\lambda_0$  – heat transfer coefficient at reference temperature  $T_0$ ;

 $C_{p0}$  – heat capacity at reference temperature  $T_0$ 



#### Mesh resolution ~2,2≈10<sup>6</sup> cells

Kolmogorov's scale  $\eta$  of smallest vortices in isotropic turbulence can be defined as  $\eta = \left(\frac{v^3}{\varepsilon}\right)^{\overline{4}}$ where v – kinematic viscosity,  $\varepsilon \approx U'^3 / L$  is energy dissipation rate, where L – system length scale, U' – turbulent pulsating component of velocity (order of magnitude less than the average velocity).

For mesh step  $h \approx 7$  cm and  $\eta \approx 5$  cm we can compute a fraction of correctly resolved turbulent energy:

$$\delta_{E_{resolved}} = \frac{\int_{k_h}^{k_\eta} E(k) dk}{\int_{k_0}^{k_\eta} E(k) dk} = \left(\frac{k_h}{k_0}\right)^{-\frac{2}{3}} \approx 0.97, \text{ where } E(k) = C\varepsilon^{2/3}k^{-5/3} \text{ is Kolmogorov's spectrum.}$$

Wavenumbers are defined through lengths:  $k_{\eta} = 2\pi / \eta$ ,  $k_0 = 2\pi / L$ ,  $k_h = 2\pi / h$ 

<u>We expect 3 % of unresolved energy</u> turbulent energy

### Instant streamlines

C

Instant streamlines, color scale – velocity (m/s), quasistationary state at  $4 \cdot 10^4$  c

 $\Delta T = 0.1$  K, Ra  $\approx 2 \cdot 10^4$  $\Delta T = 1 \text{ K}, \text{ Ra} \approx 2.10^5$  $\times 10^{-3}$ ×10<sup>-3</sup> 14 30 1 12 25 10 20 m 0 m 0 8 15 10 -1 -1 3 3 5 2 2 m -1 -1 m 0 1 m 0 m 1

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# Velocity slices

#### Streamlines and velocity magnitude

- For given temperature fluctuations gas velocity is about  $(1 \div 5) \cdot 10^{-2}$  m/s;
- Highest velocity is at the boundaries of the computational domain;
- Quasi-stationary circulation is formed along the cylinder axis between the electrode and the TPC face.



# Q-criterion & FTLE

#### Q-criterion & FTLE

 $\Delta T = 1$  K, Ra  $\approx 2 \cdot 10^5$ 

Q-criterion – vortex search:

 $Q = \frac{1}{2} \left[ \left| \Omega \right|^2 - \left| S \right|^2 \right]$ 

Difference between vorticity tensor norm  $\Omega = \frac{1}{2} \left[ \nabla \vec{v} - (\nabla \vec{v})^T \right] \text{ and viscous stress tensor norm}$   $S = \frac{1}{2} \left[ \nabla \vec{v} + (\nabla \vec{v})^T \right]$ 

Finite-time Lyapunov Exponent – coherent Lagrange structures

$$\Lambda\left(\vec{r},t_{0},\Delta t\right) = \frac{1}{|\Delta t|} \ln \sqrt{\lambda_{\max}\left[\left(\nabla \phi_{t_{0}}^{t_{0}+\Delta t}\left(\vec{r}\right)\right)^{\top} \nabla \phi_{t_{0}}^{t_{0}+\Delta t}\left(\vec{r}\right)\right]}$$

where  $t_0$  – initial time moment,

- $\Delta t$  considered period of advection,
- r coordinate vector,
- $\lambda_{max}$  maximum matrix eigenvalue operator,

 $\varphi_{t_0}$  – flowmap, matrix operator mapping coordinates of Lagrangian particle at time moment  $t_0$  to coordinates at moment  $t_0 + \Delta t$ .





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### Temperature dynamics

- Vortices occur between locally overheated and overcooled regions;
- The Nusselt number for all cases is  $10^6$ , decreasing at a  $^{-0.2}_{-0.4}$ distance of less than 0.2 metres from the walls  $^{-0.6}_{-0.8}$





# Electronic drift

When beams collide, the highest yield of elementary particles is expected at sharp angle to the central electrode. The drift velocity  $\vec{v}_d$  of electrons to the ends depends on the local temperature  $T(\vec{r})$ :

$$\vec{v}_d = \vec{v}_{d0} + K_{\Delta V/V} \cdot \vec{v}_{d0} \cdot \theta$$

where  $\vec{v}_{d0}$  is drift velocity at thermal stabilization temperature,

 $\theta$  – deviation from thermal stabilization temperature;

 $K_{\Delta V/V}$  – coefficient of linear drift velocity change under electric field 140 V/m for given thermal stabilization temperature.



Field dependence of electron drift velocity and coefficient of linear drift velocity change in  $Ar_{0.9}(CH4)_{0.1}$  for 25 °C

Peisert A., Sauli F. Drift and diffusion of electrons in gases: a compilation (with an introduction to the use of computing programs) // European organization for nuclear research. 1984. № July. P. 133.

# Electronic drift

Cauchy problem for a single electron:

$$\begin{cases} \frac{\mathrm{d}\vec{r}_e(t)}{\mathrm{d}t} = \vec{v}_d(T(\vec{r}_e))\mathrm{d}t\\ \vec{r}_e(0) = \vec{r}_{e0} \end{cases}$$

Algorithm:

1. Solution of the equations of gas dynamics and heat transfer;

2. Solution of the equations of electron motion from the track using the spatial distribution of temperature of the medium.

3. Analysis of statistical characteristics of electron drift.



Swarm of electrons drifting from the high-voltage electron to the end face of the working gas volume at the temperature fluctuation amplitude of 1 K

m/s

# Electronic drift



Distribution of deviations from mean electron drift velocity for different amplitude of temperature fluctuation

$\Delta T, \mathbf{K}$	$\Delta V_{50\%}$ , м/с	$\Delta V_{95\%}$ , м/с	$\Delta t_{50\%}, c$	$\Delta t_{95\%}, c$	$\Delta x_{50\%},  { m M}$	$\Delta x_{95\%}$ , M
0.1	0.074	0.15	3.4.10-10	6.1·10 <sup>-10</sup>	6.0·10 <sup>-6</sup>	1.2.10-5
0.25	0.08	0.41	2.9·10 <sup>-10</sup>	1.7.10-9	6.5·10 <sup>-6</sup>	3.3.10-5
1	0.22	0.82	8.9·10 <sup>-9</sup>	3.3·10 <sup>-9</sup>	1.7.10-5	6.6·10 <sup>-5</sup>
2.5	1.38	2.99	5.6·10 <sup>-9</sup>	$1.2 \cdot 10^{-8}$	1.1.10-4	2.4.10-4

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# Conclusions

- A model of heat transfer in the working gas volume of the MPD unit of the NICA accelerator complex has been developed, the corresponding initial boundary value problem has been solved by the finite element method;
- The following features of the flow were found:
  - a. heat transfer occurs primarily due to convective which is supported by  $Nu \sim 10^6$ ;
  - b. appearance of flows carrying heated and cooled gas deep into the working volume occurs due to the rotation of small vortices at the boundaries caused by temperature differences at the boundaries;
  - c. temperature deviation of  $Ar_{0.9}(CH_4)_{0.1}$  from the thermal stabilisation temperature monotonically decreases when moving away from the outer boundary of the gas volume, reaching to the centre of the working volume not more than 0.01 K;
- A method is proposed and implemented for the calculation of the drift velocity of the electron swarm in the working gas volume from the solution of the equations of motion, taking into account the dependence of the drift velocity on the local gas temperature;
- It was found that 95% of electrons have an error of no more than 7·10<sup>1</sup> μm in restoring the initial coordinate at temperature difference amplitudes of 1 K at the boundary.

The boundary conditions will be refined to account for heat generation on the ToF, and the calculations will be refined using the Nek5000 code on the supercomputer

# Thank you for your attention

# Допслайды

# Разработка экспериментальных устройств

Для запуска MPD достаточна термостабилизация 1 К, но для развития комплекса понадобится замена газа и термостабилизация до 0,1 К

- Необходимы датчики температуры высокой точности;
- Нужна система подготовки термостабилизированного газа;
- Понимание, возможен ли такой режим в принципе



Пэдовая плоскость контактирует с торцом газового объема

### Связанная система ROC



СПТЭ-2024



Термостабилизация электроники ROC установки MPD коллайдера NICA

# Проверка термостабилизации плоскости ROC

#### Эксперимент:

- Вода подается 4 раздельными линиями;
- Камера электроники ROC теплоизолирована от окружения;

 Комната кондиционирована;

 Тепловизор находится в темноте и калиброван по контактным датчикам температуры.



#### Схема эксперимента

### Используемые датчики

С точностью 0,01 К датчики Pt100 в России не сертицифируют, проведена калибровка с помощью Элемер-ТК-М150-К

https://youtu.be/\_kcda9WyVOU



Выбирались датчики из коридора ±0.02 К от среднего



https://rutube.ru/video/private/fc768c02ae8ce14493c1f14abc95949b/?p=Mdkk ueBE90\_rzah2Bd1gRA&r=plwd

#### Экспериментальная установка

# Степень термостабилизации

Внешний воздух: 24°С

SAMPA вода на входе: 25°С (стационарно)

FPGA вода на входе: 17-18 °C



Тепловизор Flir E8 был калиброван отобранным и Pt-100 терморезисторами

Сенсор	T <sub>sensor</sub> , °C	<i>T</i> <sub>camera</sub> , °C		
Желтый	21,82	21,87		
Красный	21,67	21,67		
Синий	21,50	21,56		
Серый	21,86	21,87		



Пэдовая плоскость без адаптивной температуры в контуре корпуса ROC





Вода в корпусе ROC догревается до T = 23 °C





# Газовый теплообменник

Prototype simulation parameters:

- Length 0.55 meters;
- Inlet gas is 10 K hotter than water; Gas flow is 20 200 l/min. •



- Water flow is 0.1 m<sup>3</sup>/h;



### Gas temperature sensor

- Argon flow  $\approx 800 \text{ cm}^3/\text{min}$ ;
- Water jacket is around testing tube;
- Pt100 RT + NI Controller;



Testing tube with sensor fully in



#### Sliding sensor almost out of tube

Experiment was motivated by Gleb Meshcheryakov



Time dependence of  $\Delta T$  between the control and the full-drown sliding sensor

<0.03 K gradient is achieved inside the testing tube with gas about 15 °C and room temperature 21 °C

### Gas temperature sensor



Difference between control and sliding sensor temperature on the length of wires inside gas







Prototype of gas temperature sensors with swagelok and wire support

### Выводы по экспериментальной части

- Prototypes of temperature sensors for the flowing gas have been proposed and manufactured, with an expected measurement accuracy of 0.03 K;
- Functioning front-end electronics of the ROC was stabilized using four water lines equipped with ultrasonic flowmeters and controlled via programmable logic controllers. The temperature uniformity achieved achieved on the pad plane was not worse than 0.1 K, which meets the requirements for MPD experiments.